Value Function and Optimal Trajectories for some State Constrained Control Problems

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For a given non-empty compact subset U of \mathbb{R}^k , define the set of admissible controls as:

$$\mathcal{U} := \Big\{ \mathbf{u} : (\mathbf{0}, T) o \mathbb{R}^k, ext{ measurable, } \mathbf{u}(t) \in U ext{ a.e } \Big\}.$$

➤ Consider the following control system:

(1)
$$\begin{cases} \dot{\mathbf{y}}(s) := f(\mathbf{y}(s), \mathbf{u}(s)), & \text{a.e } s \in [t, T], \\ \mathbf{y}(t) := x, \end{cases}$$

where $f: \mathbb{R}^d \times U \to \mathbb{R}^d$ is continuous, and Lipschitz continuous w.r.t x.

➤ The set of trajectories:

$$S_{[t,T]}(x) := \{ \mathbf{y}_{t,x}^{\mathbf{u}} \in W^{1,1}(t,T;\mathbb{R}^d), \ \mathbf{y}_{t,x}^{\mathbf{u}} \text{ satisfies (1) for some } \mathbf{u} \in \mathcal{U} \},$$

The multi-application: $x \rightsquigarrow S_{[t,T]}(x)$ is Lipschitz continuous.

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State constrained optimal control problems

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inf \Phi(\mathbf{y}_{t,x}^{\mathbf{u}}(T))
s.t. \mathbf{u} \in \mathcal{U}, \mathbf{y}_{t,x}^{\mathbf{u}}(s) \in \mathcal{K} \ \forall s \in [t, T].
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- \triangleright \mathcal{K} is a closed sub-set of \mathbb{R}^d .
- ▶ The final cost $Φ : \mathbb{R}^d \to \mathbb{R}$ is a Lipschitz continuous function

Outline

- Characterization of the value function under some controllability assumptions
- A general case where the controllability assumption is not satisfied
- A numerical example
- 4 End-point constrained control problem

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Set of constrained trajectories

➤ Assume that $f(x, U) := \{f(x, u), u \in U\}$ is a convex set. Then, by Filippov's theorem, the set of trajectories $S_{[t,T]}(x)$ is a compact set of $W^{1,1}([t,T])$ endowed with the C^0 -topology.

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- ➤ The set of feasible trajectories:

$$\mathcal{S}_{[t,T]}^{\mathcal{K}}(x) := \{ \boldsymbol{y} \in \mathcal{S}_{[t,T]}(x) \mid \ \boldsymbol{y}(s) \in \mathcal{K} \ \forall s \in [t,T] \}$$

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 \blacktriangleright Inward pointing (IP) condition: Assume $\overset{\overline{\circ}}{\mathcal{K}}=\mathcal{K}$ and

$$\exists \beta > 0, \quad \forall x \in \partial \mathcal{K}, \quad \min_{u \in U} f(x, u) \cdot \eta_x < -\beta.$$

Then, for $x \in \mathcal{K}, \ \mathcal{S}^{\mathcal{K}}_{[t,T]}(x) \neq \emptyset$, and $x \longmapsto \mathcal{S}^{\mathcal{K}}_{[t,T]}(x)$ is Lipschitz.

Ref: Arutyunov'84, Soner'86, Rampazzo-Vinter'99, Vinter-Frankowska'00, Clarke-Rifford-Stern'02 ...



A State constrained control problem

$$\vartheta(t,x) := \min \Big\{ \Phi(\mathbf{y}(T)) \mid \mathbf{y} \in \mathcal{S}_{[t,T]}^{\mathcal{K}}(x) \Big\},$$

- ▶ In general, ϑ is only l.s.c. on \mathcal{K} .
- ▶ Under suitable controllability assumptions, ϑ is the unique constrained viscosity solution of:

$$\partial_t \vartheta(t,x) + \mathcal{H}(x, D_x \vartheta(t,x)) = 0$$
 on $(0,T) \times \mathcal{K}$, $\vartheta(0,x) = \Phi(x)$ on \mathcal{K} .

- Under IP condition, the value function ϑ is Lipschitz continuous on \mathcal{K} . Soner'86, Motta'95, Ishii-Koike'96, Frankowska-et-al.'00-, ...
- If K is convex, f(x, u) = Ax + Bu and $\exists (\bar{x}, \bar{v}), A\bar{x} + B\bar{u} = 0$ Hermosilla-Vinter-HZ'17
- If K has a stratified structure + a local controllability assumption: Hermosilla-HZ'15, Hermosilla-Wolenski-HZ'17



How can we characterize (and compute) the value function for problems lacking controllability assumptions?

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- Characterization of the value function under some controllability assumptions.
- A general case where the controllability assumption is not satisfied
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An other alternative ...

- ightharpoonup Assume $\mathcal K$ is a closed nonempty set (no additional requirement)
- ightharpoonup Consider a function $g:\mathbb{R}^d o \mathbb{R}$, Lipschitz continuous, such that

 $\forall x \in \mathbb{R}^d$, $g(x) < 0 \Leftrightarrow x \in \mathcal{K}$.

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$$\forall x \in \mathbb{R}^d$$
, $g(x) \leq 0 \Leftrightarrow x \in \mathcal{K}$.

➤ In particular,

$$[\mathbf{y}(s) \in \mathcal{K}, \ \forall s \in [t, T]] \Longleftrightarrow \max_{s \in [t, T]} g(\mathbf{y}(s)) \le 0.$$

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➤ Assume that, for every $x \in \mathbb{R}^d$, $f(x, U) := \{f(x, u), u \in U\}$ is a convex set.

An auxiliary control problem (Altarovici-Bokanowski-HZ'13)

➤ Consider the following auxiliary control problem ($z \in \mathbb{R}$):

$$w(t,x,z) := \inf_{\mathbf{y} \in \mathcal{S}_{[t,T]}^{\mathcal{K}}(x)} \left\{ \left(\Phi(\mathbf{y}_{t,x}^{\mathbf{u}}(T)) - z \right) \bigvee \max_{s \in [t,T]} g(\mathbf{y}_{t,x}^{\mathbf{u}}(s)) \right\}.$$

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Improvement function (Mifflin'77, Solodov-Sagastizábal'04, Apkarian-et-al.'08, ...)

$$(\mathcal{P}) \qquad \min_{G(X) \le 0} F(X)$$

➤ The auxiliary optimization problem:

$$(\widehat{\mathcal{P}})$$
 $\min_{X} \left\{ (F(X) - z) \bigvee G(X) \right\}$

➤ Under Slater condition:

$$\bar{X}$$
 is optimal for $(P) \Leftrightarrow \left\{ egin{array}{l} \bar{X} \mbox{ solution of } (\widehat{\mathcal{P}}) \mbox{ for } \bar{z} = F(\bar{X}), \\ \min_{X} \left\{ (F(X) - \bar{z}) \bigvee G(X) \right\} = 0. \end{array} \right.$

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Theorem

For every $x \in \mathcal{K}$, we have:

(i)
$$\mathcal{E}$$
 $pi \vartheta(t, \cdot) = \left\{ (x, z) : w(t, x, z) \leq 0 \right\},$

(ii)
$$\vartheta(t,x) = \min \left\{ z \in \mathbb{R} , w(t,x,z) \leq 0 \right\},$$

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(iii) Under IP condition: for every $x \in \overset{\circ}{\mathcal{K}}$ we have: $\vartheta(t,x)=z, \quad s.t. \ w(t,x,z)=0.$

Define the Hamiltonian as:

$$\mathcal{H}(x,p) := \max_{u \in U} (-f(x,u) \cdot p) \quad \forall x, p \in \mathbb{R}^d.$$

Theorem

The value function w is the unique Lipschitz continuous viscosity solution of the following Hamilton-Jacobi-Bellman (HJB) equation:

$$\min\left(-\partial_t w(t,x,z) + H(x,D_x w), \ w(t,x,z) - g(x)\right) = 0 \quad [0,T[x]^d \times \mathbb{R},$$
$$w(T,x,z) = (\Phi(x) - z) \bigvee g(x), \qquad \mathbb{R}^d \times \mathbb{R}.$$

A particular choice of function g

► Let η > 0 and define the following extended set \mathcal{K}_{η} :

$$\mathcal{K}_{\eta} :\equiv \mathcal{K} + \mathbb{B}(\mathbf{0}, \eta).$$

- ▶ $g(y) := d_{\mathcal{K}}(y)$ the signed distance to \mathcal{K} .
- Consider the following auxiliary control problem :

$$w(t,x,z) := \inf_{\mathbf{y} \in S_{[t,T]}(x)} \left[\left(\Phi(\mathbf{y}(T)) - z \right) \bigvee \max_{s \in [t,T]} g(\mathbf{y}(s)) \bigwedge \eta \right],$$

where $a \wedge b = \min(a, b)$.

Theorem

Let $(t, x, z) \in [0, T] \times \mathcal{K} \times \mathbb{R}$. The following assertions hold:

(i)
$$\vartheta(t,x)-z\leq 0 \Leftrightarrow w(t,x,z)\leq 0$$
,

(ii)
$$\vartheta(t,x) = \min \Big\{ z \in \mathbb{R} , \ w(t,x,z) \leq 0 \Big\}.$$

A particular choice of function g

Theorem

The function \mathbf{w} is the unique Lipschitz continuous viscosity solution of the following HJB equation:

$$\min \left(-\partial_t w(t, x, z) + H(x, \nabla_x w), \ w(t, x, z) - g(x) \right) = 0 \quad [0, T[\times \mathcal{K}_{\eta} \times \mathbb{R}, w(T, x, z) = \psi_{\eta}(x, z), \quad \mathcal{K}_{\eta} \times \mathbb{R}, w(t, x, z) = \eta, \quad y \notin \mathcal{K}_{\eta},$$

where
$$\Psi_{\eta}(x,z) = \left[(\Phi(x) - z) \bigvee g(y) \right] \bigwedge \eta$$
.

➤ Define the exit time function:

$$\mathcal{T}(y,z) := \inf \left\{ t \in [0,T] \mid \vartheta(t,y) \le z \right\}$$
$$= \inf \left\{ t \in [0,T] \mid w(t,y,z) \le 0 \right\}$$

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- \blacktriangleright Link between ϑ and \mathcal{T} :
 - (i) \mathcal{T} is the exit time function for \mathcal{E} pi $(\Phi) \bigcap (\mathcal{K} \times \mathbb{R}^d)$,
 - (ii) $T(y,z)=t \Rightarrow w(t,y,z)=0$,
 - (iii) $\vartheta(t, y) = \inf\{z \mid \mathcal{T}(y, z) \leq t\}.$

Reconstruction of optimal trajectories

Proposition

Let $x \in \mathcal{K}$ such that $\vartheta(t, x) < \infty$. Define $z^* := \vartheta(t, x)$.

• Let $(\mathbf{y}^*, \mathbf{z}^*)$ be the optimal trajectory for the auxiliary control problem associated with the initial point $(\mathbf{x}, \mathbf{z}^*) \in \mathcal{K} \times \mathbb{R}$. Then, the trajectory \mathbf{y}^* is optimal for the original control problem. The converse is also true.

Reconstruction of optimal trajectories - Algorithm A.

- For $n \ge 1$, consider $(t_0 = 0, t_1, ..., t_{n-1}, t_n = T)$ a uniform partition of [0, T] with $\Delta t = \frac{T}{n}$.
- ► Let $\{\mathbf{y}^n(\cdot), \mathbf{z}^n(\cdot)\}$ be a trajectory defined recursively on the intervals $(t_{i-1}, t_i]$, with $\mathbf{z}^n(\cdot) := z = \vartheta(0, y)$ and $\mathbf{y}^n(0) = y$.

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- ► [Step 1] Knowing $y_k^n = \mathbf{y}^n(t_k)$, choose the optimal control at t_k s.t.:

$$u_k^n \in \arg\min_{u \in U} \left(w(t_k, y_k^n + \Delta t f_{\Delta t}(y_k^n, u), z) \right).$$

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► [Step 2] Define $\mathbf{u}^n(t) := u_k^n$, $\forall t \in (t_k, t_{k+1}]$ and $\mathbf{y}^n(t)$ on $(t_k, t_{k+1}]$ as the solution of

$$\dot{\mathbf{y}}(t) := f(\mathbf{y}(t), \mathbf{u}^n(t)) \text{ a.e } t \in (t_k, t_{k+1}],$$

with initial condition $\mathbf{y}^n(t_k)$ at t_k and $\mathbf{z}^n(\cdot) := \mathbf{z}$.

Let $\{\mathbf{y}^n(\cdot), \mathbf{z}^n(\cdot), \mathbf{u}^n(\cdot)\}$ be a sequence generated by algorithm A for $n \geq 1$. Then, the sequence of trajectories $\{\mathbf{y}^n(\cdot)\}_n$ has cluster points with respect to the uniform convergence topology. For any cluster point $\bar{\mathbf{y}}(\cdot)$ there exists a control law $\bar{\mathbf{u}}(\cdot)$ such that $(\bar{\mathbf{y}}(\cdot), \bar{\mathbf{z}}(\cdot), \bar{\mathbf{u}}(\cdot))$ is optimal for the auxiliary control problem.

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► Let \mathbf{w}^{\triangle} be a numerical approximate solution such that,

$$|w^{\Delta}(t,y,z)-w(t,y,z)|\leq E_1(\Delta t,\Delta y),$$

where $E_1(\Delta t, \Delta y) \to 0$ as $\Delta t, \Delta y \to 0$.

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► Let $\{Y^n(.), \mathbf{u}^n(.)\}$ be the sequence generated by the algorithm A with \mathbf{w}^{Δ} .

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where $E_1(\Delta t, \Delta y) \to 0$ as $\Delta t, \Delta y \to 0$.

- ► Let $\{Y^n(.), u^n(.)\}$ be the sequence generated by the algorithm A with w^{\triangle} .
- ightharpoonup Then, $(Y^n)_n$ converges to an optimal trajectory for the auxiliary control problem.

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Abort landing problem in presence of windshear

(Miele, Wang and Melvin(1987,1988); Bulirsch, Montrone and Pesch (1991..); Botkin-Turova(2012 ...))

Consider the flight motion of an aircraft in a vertical plane:

$$\begin{cases} \dot{x} = V\cos\gamma + w_x \\ \dot{h} = V\sin\gamma + w_h \\ \dot{V} = \frac{F_T}{m}\cos(\alpha + \delta) - \frac{F_D}{m} - g\sin\gamma - (\dot{w}_x\cos\gamma + \dot{w}_h\sin\gamma) \\ \dot{\gamma} = \frac{1}{V}(\frac{F_T}{m}\sin(\alpha + \delta) + \frac{F_L}{m} - g\cos\gamma + (\dot{w}_x\sin\gamma - \dot{w}_h\cos\gamma)) \end{cases}$$

where

$$\dot{w}_{x} = \frac{\partial w_{x}}{\partial x} (V \cos \gamma + w_{x}) + \frac{\partial w_{x}}{\partial h} (V \sin \gamma + w_{h})$$

$$\dot{w}_{h} = \frac{\partial w_{h}}{\partial x} (V \cos \gamma + w_{x}) + \frac{\partial w_{h}}{\partial h} (V \sin \gamma + w_{h})$$

and

- $F_T := F_T(V)$ is the thrust force
- $F_D := F_D(V, \alpha)$ and $F_L := F_L(V, \alpha)$ are the drag and lift forces
- $w_x := w_x(x)$ and $w_h := w_h(x, h)$ are the wind components
- m, g, and δ are constants.

Controlled system

- Consider the state $\mathbf{y}(.) = (\mathbf{x}(.), \mathbf{h}(.), \mathbf{V}(.), \gamma(.), \alpha(.))$.
- The control variable \mathbf{u} is the angular speed of the angle of attack α .
- Let T be a fixed time horizon and let \mathcal{U} be the set of admissible controls

$$\mathcal{U} := \left\{ \mathbf{u} : (0,T)
ightarrow \mathbb{R}, ext{ measurable, } \mathbf{u}(t) \in U ext{ a.e }
ight\}$$

where *U* is a compact set.

• The controlled dynamics in this case is:

$$\begin{cases} \dot{x} = V \cos \gamma + w_x, \\ \dot{h} = V \sin \gamma + w_h, \\ \dot{V} = \frac{F_T}{m} \cos(\alpha + \delta) - \frac{F_D}{m} - g \sin \gamma - (\dot{w}_x \cos \gamma + \dot{w}_h \sin \gamma), \\ \dot{\gamma} = \frac{1}{V} (\frac{F_T}{m} \sin(\alpha + \delta) + \frac{F_L}{m} - g \cos \gamma + (\dot{w}_x \sin \gamma - \dot{w}_h \cos \gamma)), \\ \dot{\alpha} = \mathbf{u}. \end{cases}$$

Formulation of the optimal control problem

• Aim: Maximize the minimal altitude over a time interval:

$$\min_{\theta \in [0,t]} h(\theta)$$

while the aircraft stays in a given domain \mathcal{K} .

Formulation of the optimal control problem

• Aim: Maximize the minimal altitude over a time interval:

$$\min_{\theta \in [0,t]} h(\theta)$$

while the aircraft stays in a given domain K.

Consider the following optimal control problem:

$$(\mathbf{P}): \quad \vartheta(t,y) = \inf \left\{ \max_{\theta \in [0,t]} \Phi(\mathbf{y}^{\mathbf{u}}_{y}(\theta)), \left| \mathbf{u} \in \mathcal{U}, \text{ and } \mathbf{y}^{\mathbf{u}}_{y}(s) \in \mathcal{K}, \ \forall s \in [0,t] \right. \right\}$$

where $\Phi(\mathbf{y}_y^y(.)) = H_r - h(.)$, H_r being a reference altitude, and \mathcal{K} is a set of state constraints.

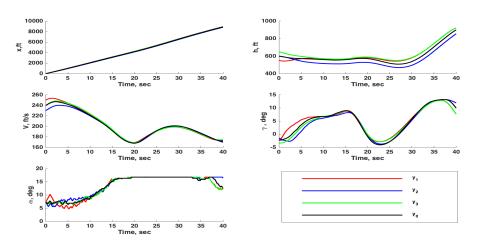


Figure: Optimal trajectories for different initial conditions

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➤ Consider the end-point constrained control problem

➤ An associated auxiliary control problem ($z \in \mathbb{R}$) can be defined as:

$$w(t,x,z) := \inf_{\mathbf{u} \in \mathcal{U}} \left\{ \left(\Phi(\mathbf{y}_{t,x}^{\mathbf{u}}(T)) - z \right) \bigvee g(\mathbf{y}_{t,x}^{\mathbf{u}}(T)) \right\}.$$

 \blacktriangleright Assume that $\Phi: \mathbb{R}^d \to \mathbb{R}$ and $g: \mathbb{R}^d \to \mathbb{R}$ are of C^1

- ► Let $x_o \in \mathbb{R}^d$ and let $z^* := \min\{z : w(0, x_o, z) \le 0\}$.
- ▶ If $z^* < \infty$, then there exists $\mathbf{u}^* \in \mathcal{U}$ and its associated trajectory $\mathbf{y}^* \in \mathcal{S}_{[0,T]}(x_o)$ such that:

$$g(\mathbf{y}^*(T)) \leq 0, \quad \vartheta(0, x_o) = z^* = \Phi(\mathbf{y}^*(T)).$$

➤ Denote $H(x, u, p) = \langle p, f(x, u) \rangle$. There exists $(\mathbf{p}_x^*, \mathbf{p}_z^*)$ satisfying:

$$\begin{array}{rcl}
-\dot{\mathbf{p}}_{x}^{*}(s) & = & \partial_{x}H(\mathbf{y}^{*}(s),\mathbf{u}^{*}(s),\mathbf{p}_{x}^{*}(s)) \\
-\dot{\mathbf{p}}_{z}^{*}(s) & = & 0 \\
\begin{pmatrix}
\mathbf{p}_{x}^{*}(T) \\
\mathbf{p}_{z}^{*}(T)
\end{pmatrix} & \in & \partial_{x,z}\left\{(\Phi(\mathbf{y}^{*}(T))-z^{*})\bigvee g(\mathbf{y}^{*}(T))\right\}
\end{array}$$

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-\dot{\mathbf{p}}_{z}^{*}(s) & = & 0 \\
\begin{pmatrix}
\mathbf{p}_{x}^{*}(T) \\
\mathbf{p}_{z}^{*}(T)
\end{pmatrix} & = & \begin{pmatrix}
\lambda_{o}\nabla\Phi(\mathbf{y}^{*}(T)) + \lambda\nabla g(\mathbf{y}^{*}(T)) \\
-\lambda_{o}
\end{pmatrix}$$

where $\lambda_o, \lambda \in [0, 1]$ and $\lambda_0 + \lambda = 1$.

▶ Denote $H(x, u, p) = \langle p, f(x, u) \rangle$. There exists $(\mathbf{p}_x^*, \mathbf{p}_z^*)$ satisfying:

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\begin{pmatrix}
\mathbf{p}_{x}^{*}(T) \\
\mathbf{p}_{z}^{*}(T)
\end{pmatrix} & = & \begin{pmatrix}
\lambda_{o}\nabla\Phi(\mathbf{y}^{*}(T)) + \lambda\nabla g(\mathbf{y}^{*}(T)) \\
-\lambda_{o}
\end{pmatrix}$$

where $\lambda_o, \lambda \in [0, 1]$ and $\lambda_0 + \lambda = 1$.

➤ For a.e $s \in [0, T]$,

$$\mathcal{H}(\mathbf{y}^*(s),\mathbf{p}_x^*(s))=H(\mathbf{y}^*(s),\mathbf{u}^*(s),\mathbf{p}_x^*(s)).$$

▶ Denote $H(x, u, p) = \langle p, f(x, u) \rangle$. There exists $(\mathbf{p}_x^*, \mathbf{p}_z^*)$ satisfying:

$$\begin{array}{rcl}
-\dot{\mathbf{p}}_{x}^{*}(s) & = & \partial_{x}H(\mathbf{y}^{*}(s),\mathbf{u}^{*}(s),\mathbf{p}_{x}^{*}(s)) \\
-\dot{\mathbf{p}}_{z}^{*}(s) & = & 0 \\
\begin{pmatrix}
\mathbf{p}_{x}^{*}(T) \\
\mathbf{p}_{z}^{*}(T)
\end{pmatrix} & = & \begin{pmatrix}
\lambda_{o}\nabla\Phi(\mathbf{y}^{*}(T)) + \lambda\nabla g(\mathbf{y}^{*}(T)) \\
-\lambda_{o}
\end{pmatrix}$$

where $\lambda_o, \lambda \in [0, 1]$ and $\lambda_0 + \lambda = 1$.

➤ For a.e $s \in [0, T]$,

$$\mathcal{H}(\mathbf{y}^*(s),\mathbf{p}_x^*(s)) = H(\mathbf{y}^*(s),\mathbf{u}^*(s),\mathbf{p}_x^*(s)).$$

➤ Moreover,

$$(\mathbf{p}_{x}^{*}(s),\mathbf{p}_{z}^{*}(s))\in\partial_{x,z}w(s,\mathbf{y}^{*}(s),z^{*}).$$

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- ➤ The same framework can be extended to:
 - final state constraints and time-dependent state constraints
 - impulsive control problems (Forcadel-Rao-HZ'13)
 - supremum running cost problems (Assellaou-Bokanowski-Desilles-HZ'17),
 - stochastic control setting (Bokanowski-Picarelli-HZ'16)
- ➤ The relationship between the PMP and the auxiliary value function can be (easily) derived for control problems with a finite number of state constraints.

....thanks for your attention.